# **isoo/izoo MERSIAL MERSUREMENT MEASUREMENT UNIT**

Installation and Environmental Manual





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# **PRODUCT** OVERVIEW

The HGuide i300/i200 is a high performance Micro-Electro-Mechanical System (MEMS) based Inertial Measurement Unit (IMU) designed to meet the needs of applications across various markets including agriculture, AUVs, industrial equipment, robotics, survey/mapping, stabilized platforms, transportation, UAVs, and UGVs. With industry standard communication interfaces and a wide input voltage range the HGuide i300/i200 is easily integrated into a variety of architectures. The extremely small size, lightweight, and low power make the HGuide i300/i200 ideal for many applications.

The HGuide i300/i200 includes MEMS gyroscopes and accelerometers. In addition, the HGuide i300/i200 employs an internal environmental isolation system to attenuate unwanted inputs commonly encountered in real world applications. The internal isolation and other proprietary design features ensure the HGuide i300/i200 is rugged enough to meet the needs of the most demanding users.

The HGuide i300/i200 is both hardware and software compatible with the HG4930 IMU. It is also software compatible with the HG1120 IMU with their message descriptions contained in those device manuals.

The HGuide i300/i200 is not ITAR controlled. Its Export Control Classification Number (ECCN) is 7A994.

For more information, email <u>hguide.sales@honey well.com</u> or contact us on our website aerospace.honeywell.com/I300

### HGUIDE DATA READER/ INTEGRATION

The Honeywell HGUIDE DATA READER is a web deployed software integration tool which can configure the i300/i200 for message types and baud rate. The software tool also provides real time and "Off Device" integration support.

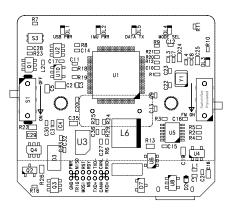
The software integration tool will display and record data, generates supporting message documentation, and includes an example Windows executable which will parse and log data. The program will also export data to CSV format for easy plotting.

The Honeywell HGuide Data reader provides a software development kit (SDK) including C/C++ source code, header files, DLL, and essential functions. See "Bit Stream" window to produce the SDK.

An evaluation kit is also available for separate purchase. Connect the evaluation board to the IMU being careful to align the pins to the connector.

Connect a micro-USB cable. Make sure the switch is on. Verify that both green LEDs power on. Once data transmission starts the orange LED will turn on. Data can be monitored using any terminal program or the data reader program. The Windows Device Manager should show a new port.

If using the Honeywell Data reader, be sure to press the "scan/hunt" button on the introductory screen. The program will automatically do an initial search but will time out if device not connected.



CAN ONLY Replace 120 Ohm Resistor Between CANH and CANL.

Evaluation kit (68009732-003) with shown board is available for purchase.

#### EVAL KIT BOARD MOUNTING TO i300/i200

Use two 2-56 x 3/16" SHCS Grainger part number 1GU11. Driver size is 5/64" **RECOMMENDED i300 MOUNTING** Use two 4-40 x 3/8" SHCS (M3.0 x 0.5) screws. Grainger part number 5UGX4. Driver size is 3/32"

HGuide Data Reader		-	
HGuide Data F	leader		i300 Setup Record Play File Export
Graphics Text Bi	t Stream		
Health 🔬 Control 🖄 Navigation 🗹 Co	ntrol Plots 🦯 Navigation P	lots	Snaps
Device Summary Status		Contro	
Group Name Status Oth	er Device Data	Value	Angular rate Acceleration 0.01 10.11
Gyro Guid Magnetometer Dev: System Status Sor	rol Frame Rate ance Frame Rate Lee Temperature Ftware Version Device ID Performace Grade	1 1 35.93 °C 11 12 10	CO1 10.11 (rad/s) (rws/s2)
itatus Details Name	Group	Status	
Gyro Bit Summary Accelerator Bit Summary Nag Bit Summary CBIT Summary Gyro Statistics Summary Accelerometer Statistics Summary Accelerometer Statistics Summary Magnetometer Temperature Summary Normal Mode Primary CRC Normal Mode Primary CRC Normal Mode Secondary CRC Factory Configuration CRC Factory Configuration CRC IO Configuration CRC Primary Image Boot	Gyro Accelerometer Magnetometer System Status Gyro Accelerometer Accelerometer Magnetometer Magnetometer System Status System Status System Status System Status System Status		Angular cates Acceleration 0.00 (radio) 0.03 (radio) 0.0
Memory Bit Summary	System Status YNC ONLY Frequen	acy: 600/100	Start Time: 15:53:46 Elapsed Time: 00:03:5

# **PIN OUTS/POWER** SIGNAL LIST

Typical power draw is 500 mW. The i300/i200 will work with most PC USB ports; however, Honeywell recommends a non-PC power supply for permanent installations.

This device has been designed to meet stringent EMI and EMC requirements, and as such, the user should shield the I/O cabling and provide chassis ground connection to the IMU housing.

	i300 14 PIN CONNECTOR J1							
Pin#	Signal Name	Input/Output & Signal type	Signal Function					
1	Ground	Ground	Ground					
2	Power	Power	+4.75 to 36 VDC					
3	SCLK <sup>1</sup>	Input, +3.3 to 5.5 VDC	Serial Clock Output from Master					
4	COM1 TX / MISO <sup>1</sup>	Output, +5 Volt TTL	Asynchronous or SPI Output					
5	COM1 RX / MOSI <sup>1</sup>	Input - +3.3 to 5.5 VDC	Asynchronous or SPI Input					
6	RESET	Input - +3.3 to 5.5 VDC	Device reset input discrete. CMOS Compatible. Device will remain in reset while at logic 0.					
7	Slave Select <sup>1</sup>	Input, +3.3 to 5.5 VDC	Slave Select (Active Low)					
8	System TOV Mark	Output - 5 Volt BiCMOS	Data Ready on Rising Edge to 5 VDC. Rise and Fall Time < 50 Nanoseconds.					
9	COM3_TX_H		Asynchronous High					
10	COM1_TX_L	Output RS-422	Asynchronous Low					
11	COM2 / CAN_H	Bi-directional ISO 11898-2	Asynchronous High					
12	COM2 / CAN_L	120 Ohm Termination Resistor Required	Asynchronous Low					
13	COM3_RX_H	Input RS-422	Asynchronous High					
14	COM1_RX_L	input K3-422	Asynchronous Low					

1 Contact Honeywell for future SPI product availability.

## **DATA READY** SIGNAL DESCRIPTION

<b> </b> ◀──		1/3600 Sec	
Sensor Sampling @ 3600Hz	Processing 65 - 80 uS	Data Ready & TX Start	DATA RDY FALLING EDGE

Data Ready Rising and Falling at Message Frame Rate. Data transmission will continue across additional 3600 Hz frames dependent upon selected bit rate.

The Data Ready falling edge can be used to provide a time mark to a data recording system which will provide time of validity for recorded IMU data. This is often GPS time. Some customers also need to know when Honeywell data transmission starts so that the data can be immediately processed or manage a data bus.

### HGUIDE i300/i200 PERFORMANCE

	GYRO ROOM TEMPERATURE TYPICAL PERFORMANCE								
Distributor Ordering Part Number	Performance Class	Bias Repeatability (°/hr 1ơ)	Bias In-run Stability (°/hr 1ơ)	ARW (°∕√hr)	Scale Factor Repeatability (ppm 10)				
i300	"B"	65	3	0.15	1400				
i300	"A"	90	5	0.25	1400				
i200	"C"	260	10	0.30	2500				

ACCELEROMETER ROOM TEMPERATURE TYPICAL PERFORMANCE								
Distributor Ordering Part Number	Performance Class	Bias Repeatability (mg 1ơ)	Bias In-run Stability (mg 1ơ)	VRW (m/s/√hr)	Scale Factor Repeatability (ppm 1ơ) <sup>1</sup>			
i300	"B'	1	0.02	0.02	250 <sup>1</sup>			
i300	"A"	2	0.03	0.03	500 <sup>1</sup>			
i200	"C"	5	0.06	0.04	10001			

1 Model as a +1500 PPM offset with the one sigma value as shown.

GYRO AND ACCELEROMETER ROOM TEMPERATURE TYPICAL NOISE @ 400 HZ BW							
Distributor	Axes	Control Data (1σ) @ .	1800 Hz Data Rate	Control Data (1ơ) @ 1800 Hz Data Rate			
Ordering Part Number		Gyro (Deg/s)	Accel (mg)	Gyro (Deg)	Accel (mm/s)		
i300	Х	0.15	5	0.0005	0.12		
i300	Y/Z	0.15	2.5	0.0005	0.06		
i200	All Axes	0.15	8	0.05	0.30		

### USING HONEYWELL DATA

#### Controlling (Messages 0xA1 or 0xAC)

Honeywell provides high bandwidth (400 Hz information) at 1800 Hz. General "rule of thumb" is that sensor control bandwidth should be 5 times the structure being controlled – allowing control of an 80 Hz device. If you are controlling lower frequency platforms (like a car) – filter the 1800 Hz data to the desired bandwidth.

#### Navigation (Messages 0xA3 & 0xAE)

Navigating requires that single integration of angular rates to attitude and the double integration of acceleration into position. This navigation format provides data relative to the prior frame and often referred to as Delta Velocity / Theta or Incremental Velocity / Angles. The data is directly integrable in that the data is not "per second" but rather per the length of the navigation frame (100 Hz, 200 Hz or 300 Hz).

#### Interleaved (Multiple Messages & CAN)

Get both Control and Navigation data by selecting a message which sends out data on multiple frame rates. This is the default data for the i300/i200 because it demonstrates all sensor types. If you are one of the few who need this – study carefully the right columns on the Interleave table. For the rest – use the Honeywell HGuide Data Reader to send out just Control or Navigation data.

# **PRODUCT** DEFAULT MESSAGE

The HGuide i300 / i200 Inertial Measurement Units (IMUs) are factory configured with 0xA1/0xA2 Interleaved Messages at 921.6 KBaud. Use the HGuide Data Reader (HGDR) to change the default message to alternates.

### DEFAULT i300 / i200 MESSAGE DEFAULT BAUD RATE FROM FACTORY IS 921.6 KBAUD

Message Name	Message Information		Data Ra	ites (Hz)	Available KBaud Rates			
	Control	Nav	Control	Nav	460.8	921.6	1000	
0xA1 & A2	Х		600	100	Х	Х	Х	
Interleaved		Х	1200	200		Х	Х	
			1800	300		Х	х	

DEFAULT i300/i200 MESSAGE INTERLEAVE TABLE										
Message Name	Mes	Message		Data Rates (Hz)		Interleave Transmission Sequence				
	Control	Nav	Control	Nav	1	2	3	4	5	6
0xA1 & 0xA2 Interleaved			600	100						
	Х	Х	1200	200	0xA2	0xA1	0xA1	0xA1	0xA1	0xA1
			1800	300						

### **ALTERNATE MESSAGE** SELECTION

ALTERNATE MESSAGES. USE THE HGUIDE DATA READER TO SELECT MESSAGES AND BAUD RATES											
Allowed Messages	N	lessage	Informa	ition	Data Rates (Hz)		Available KBaud Rates				
	Control	Mag	Nav	Descrip- tion	Control	Nav	115.2	230.4	460.8	921.6	1000
CAN	Х	Х	Х	ALL	600	100		Х	Х	Х	Х
0x01	Х			HG4930	600	NA			Х	Х	Х
0x01 & 0x02 Interleaved	Х		Х	HG4930	600	100			Х	Х	Х
0x04	Х			HG1120	1800	NA			Х	Х	Х
0x04 & 0x05 Interleaved	Х		Х	HG1120	1800	300			Х	Х	Х
0x0C	Х			HG1120	600	NA			Х	Х	Х
0x0C & 0x0D Interleaved	Х		Х	HG1120	600	100				Х	Х
0xA1	Х				600				Х	Х	Х
				i300/i200	1200	NA			Х	Х	Х
					1800				Х	Х	Х
0xA3			Х			100	Х		Х	Х	Х
				i300/i200	NA	200	Х		Х	Х	Х
						300			Х	Х	Х
OxAC	Х	Х			600				Х	Х	Х
				i300/i200	1200	NA			Х	Х	Х
					1800					Х	Х
OxAC & OxAD	Х	Х	Х		600	100			Х	Х	Х
Interleaved				i300/i200	1200	200				Х	Х
					1800	300				Х	Х
OxAE		Х	х			100	Х		Х	Х	Х
				i300/i200	NA	200	Х		Х	Х	Х
						300	Х		Х	Х	Х

For Legacy HG1120 and HG4930 messages, consult their respective manuals for message definitions.

Use the HGuide Data Reader (HGDR) to configure messages and baud rates. CAN messages may be enabled via the HGDR but require configuration upon each power cycle.

### **INTERLEAVE TABLE** ALTERNATE MESSAGES

INTERLEAVE	INTERLEAVE TABLE. TRANSMISSION SEQUENCE IS AT THE CONTROL RATE											
Allowed Messages	м	Message Information			Data Rates (Hz)		Interleave Transmission Sequence @Control Rate					
	Control	Mag	Nav	Device	Control	Nav	1	2	3	4	5	6
CAN	х	Х	х	ALL	600	100	C1 C2 M1 I1 I2 I3	C1 C2 M1	C1 C2 M1	C1 C2 M1	C1 C2 M1	C1 C2 M1
0x01 & 0x02 Interleaved	Х		х	HG4930	600	100	0x02	0x01	0x01	0x01	0x01	0x01
0x04 & 0x05 Interleaved	Х		х	HG1120	1800	300	0x05	0x04	0x04	0x04	0x04	0x04
0x0C & 0x0D Interleaved	Х		Х	HG1120	600	100	0x0D	0x0C	0x0C	0x0C	0x0C	0x0C
					600	100						
OxA1 & OxA2 Interleaved	х		Х	i300/ i200	1200	200	0xA2	0xA1	0xA1	0xA1	0xA1	0xA1
					1800	300						
					600	100						
OxAC & OxAD Interleaved	Х	х	X	i300/ i200	1200	200	OxAD	0xAC	OxAC	OxAC	0xAC	0xAC
					1800	300						

### ASYNCHRONOUS MESSAGE EXAMPLE

	ASYNCHRONOUS MESSAGE								
Message Detail Template OxAD Control, Mag, Status & Delta Data Apply the LSB, Byte, and Units information to all other message ID's									
Position	Parameter	Description	Bytes	LSB Weight	Units/LSB				
1	Address	0x0E	1	NA					
2	Message ID	OxAD	1	NA					
3		Angular Rate X	2						
4		Angular Rate Y	2	2-11	rad/sec				
5	Control	Angular Rate Z	2						
6	Control	Linear Acceleration X	2						
7		Linear Acceleration Y	2	0.3048*2-5	m/sec <sup>2</sup>				
8		Linear Acceleration Z	2						
9		Mag Field X 2							
10	Mag	Mag Field Y	2	0.438404	Milli-Gauss				
11		Mag Field Z	2						
12	Status	Status Word 1	2	See Table	See Table				
13	Status	Status Word 2	2	NA	Reserved				
14		Delta Angle X	4		radians				
15		Delta Angle Y	4	<b>2</b> <sup>-33</sup>	or equivalently, radians/second/Hz				
16	Navigation	Delta Angle Z	4						
17	Navigation	Delta Velocity X	4		m/sec				
18		Delta Velocity Y	4	0.3048*2-27	or equivalently,				
19		Delta Velocity Z	4		m/sec²/Hz				
20	Checksum	Checksum	2	NA	Total of 50 Bytes				

All messages are a subset of this message set.

Apply the LSB, Byte, and Units information to all other Message ID's.

The LS byte and 16-bit word are first (little endian).

Asynchronous Communications are 8 Bits, One Start Bit, One Stop Bit.

See HGDR SDK (in BIT STREAM tab)

Select if the HGuide SDK Documentation and/or Parser Files are to be generated

HGuide SDK / Message Protocal Definitions in HTML Documentation

Parser Files in... CPP 
Generate Print Methods

KEXTact Demo EXE/DLL

OK Cancel

### **STATUS AND CHECKSUM** DESCRIPTIONS

	i300/i200 STATUS WORD							
Bit	Definition	Values						
0–3	4-bit Counter	0-15						
4–7		O (No Active Output)						
	Control Data Output	1 (600 Hz)						
	Controt Data Output	2 (1200 Hz)						
		3 (1800 Hz)						
8-11		O (No Active Output)						
	Navigation Data Output	1 (100 Hz)						
	Navigation Data Output	2 (200 Hz)						
		3 (300 Hz)						
12–15	BIT (Gyro/Accel/Mag/Summary)	Logic 0 is Pass						

The Checksum is the sum of all message data (positions 1 ... 19 of example message), taken as 16 bit words, and summed without regard for rollover.

This pseudo code illustrates the checksum algorithm for the 0xA1 message:		
u16sum = 0;		
for (i=0; i<9; i++) // (20-2)/2=9		
{		
Checksum = u16_msg_array[9];		
if (Checksum != u16sum) {checksum error}		

The HGuide Data Reader with its associated software development tools provide real time examples of checksum calculations.

### ASYNCHRONOUS MESSAGES ABBREVIATED

OXA1 CONTROL DATA		
Position	Parameter Description	Bytes
1	IMU Address - 0x0E	1
2	Message ID - 0xA1	1
3-8	Control Data	12
9-10	Status	4
11	Checksum	2
	Total Bytes	20

<b>OXAC CONTROL &amp; MAG DATA</b>		
Position	Parameter Description	Bytes
1	IMU Address - 0x0E	1
2	Message ID - 0xAC	1
3-8	Control Data	12
9-11	Mag Data	6
12-13	Status	4
14	Checksum	2
	Total Bytes	26

<b>0XA2 CONTROL &amp; NAVIGATION DATA</b>		
Position	Parameter Description	Bytes
1	IMU Address - 0x0E	1
2	Message ID - 0xA2	1
3-8	Control Data	12
9-10	Status	4
11-16	Navigation Data	24
17	Checksum	2
	Total Bytes	44

OXAD CONTROL, NAVIGATION & MAG DATA		
Position	Parameter Description	Bytes
1	IMU Address - 0x0E	1
2	Message ID - 0xAD	1
3-8	Control Data	12
9-11	Mag Data	6
12-13	Status Word	4
14-19	Navigation Data	24
20	Checksum	2
	Total Bytes	50

OXA3 NAVIGATION DATA		
Position	Parameter Description	Bytes
1	IMU Address - 0x0E	1
2	Message ID - 0xA3	1
3-8	Navigation Data	24
9-10	Status	4
11	Checksum	2
	Total Bytes	32

<b>OXAE NAVIGATION &amp; MAG DATA</b>		
Position	Parameter Description	Bytes
1	IMU Address - 0x0E	1
2	Message ID - 0xAE	1
3-8	Navigation Data	24
9-11	Mag Data	6
12-13	Status	4
14	Checksum	2
	Total Bytes	38

See 0xAD Combined Control & Inertial for detailed contents.

See example 0xAD Combined Control & Inertial Message Table in this document or the HGDR SDK (in BIT STREAM tab). Select if the HGuide SDK Documentation and/or Parser Files are to be generated

HGuide SDK / Message Protocal Definitions in HTML Documentation

Parser Files in... CPP 
Generate Print Methods

K Cancel

OK Cancel

### **CAN MESSAGES** ABBREVIATED

C1 CONTROL DATA		
#	Parameter	Bytes
NA	Arbitration ID	NA
1	Angular Rate X	2
2	Angular Rate Y	2
3	Angular Rate Z	2
4	Status Word	2

C3 MAGNETIC DATA		
#	Parameter	Bytes
NA	Arbitration ID	NA
1	Mag X	2
2	Mag Y	2
3	Mag Z	2

C2 CONTROL DATA		
#	Parameter	Bytes
NA	Arbitration ID	NA
1	Linear Acceleration X	2
2	Linear Acceleration Y	2
3	Linear Acceleration Z	2
4	Reserved	2

I1 NAVIGATION DATA		
#	Parameter	Bytes
NA	Arbitration ID	NA
1	Delta Angle X	4
2	Delta Velocity X	4

12 NAVIGATION DATA		
#	Parameter	Bytes
NA	Arbitration ID	NA
1	Delta Angle Y	4
2	Delta Velocity Y	4

I3 NAVIGATION DATA#ParameterBytesNAArbitration IDNA1Delta Angle Z42Delta Velocity Z4

See OxAD Combined Control & Inertial for LSB weights and units.

CAN ID TABLE					
(0xC, 0xD, 0xA1, 0xA2)					
Message	Packet	CAN-11	CAN-29		
0xC	C1	0x121	0x04924921		
0xC	C2	0x122	0x04924922		
0xC	C3	0x126	0x04924926		
0xD	11	0x123	0x04924923		
0xD	12	0x124	0x04924924		
0xD	13	0x125	0x04924925		
0xA1	C1	0x141	0x04924941		
0xA1	C2	0x142	0x04924942		
0xA1	C3	0x146	0x04924946		
0xA2	11	0x143	0x04924943		
0xA2	12	0x144	0x04924944		
0xA2	13	0x145	0x04924945		

CAN ID TABLE				
(0x01,0x02,0xAC,0xAD)				
0x01	C1	0x131	0x04924931	
0x01	C2	0x132	0x04924932	
0x02	11	0x133	0x04924933	
0x02	12	0x134	0x04924934	
0x02	13	0x135	0x04924935	
OxAC	C1	0x151	0x04924951	
OxAC	C1	0x152	0x04924952	
OxAC	C3	0x156	0x04924956	
OxAD	11	0x153	0x04924953	
OxAD	12	0x154	0x04924954	
OxAD	13	0x155	0x04924955	

Use CAN ID's to design DBC Files.

# SPI PROTOCOL

# The SPI PROTOCOL will be available in future product releases. Contact Honeywell for availability.

The messages are identical in content to the asynchronous Control/Inertial messages except that Position 0 will be added and contain a 1 byte field containing the number of bytes of data (not including spare bytes) in the message.

The External SPI device will be coming in asynchronous to the Control/Inertial message sequence. Each SPI message in the Control/Inertial set will be a constant length. The Control message will have spare bytes at the end, NOT included in the checksum, to match the length of the Inertial Message.

The SPI clock frequency must be at least 2 MHz or no faster than 9 Mhz. The SPI clock polarity and phase are set to one (1). SPI data order is MSB first. A 4-wire SPI implementation is used.

The DATA\_RDY signal must be used to synchronize your application to the data being produced to ensure a consistent data set. The DATA\_RDY signal must trigger an SPI fetch, and the clock rate must be fast enough to fetch an entire message within the Control data rate (either 1800 or 600 Hz).

The SPI\_SS signal should be set, then the application should clock 408 (51\*8) SPI bits before resetting the SPI\_SS signal.

### **ENVIRONMENTAL/** COMPLIANCE

ENVIRONMENTAL AND COMPLIANCE INFORMATION			
ltem	Operating	Non-Operating	Units
Temperature	-54 to +85 -40 to +85 (Full Performance)	-55 to +95	°C
Temperature Shock	±3 Operating ±0.8 Full Performance	-40 to + 85 in 15 Minutes Measure on Top of Device with Thermocouple	°C/minute
Random Vibration	5 g's RMS	12 g's RMS	NA
Shock	15 g bump half-sine, 6 ms duration, both polarities, each axis, per IEC 60068-2-27	40 g Shock at 11 msec duration per MIL-STD-810G Method 516.7 Procedure I 500 g's 0.5 mSec, Half Sine	NA
Static Acceleration	> 250 g's of static acceleration in all directions and recover within 25 milliseconds		NA
Altitude	0 to 12000, Mean Sea Level		Meters
Magnetic Field	±10	No Known Sensitivity	Gauss
Acoustic Rectification	147 dB, SPL, 20 - 8000 Hz	No Known Sensitivity	NA
Regulatory	DS/EN 13309:2010 & ISO 13766-1:2018 (Construction Machinery), NA EN ISO 14982:2009 (Agricultural and Forestry Machinery)		
Materials	RoHS Compliant and RoHS Process Compatible NA		NA
WEEE Compliance	Classified as electrical and electronic equipment. Xertical Must be sent to separate collection facilities for recovery and recycling.		X
Packaging	IP68 Compliant		

# EMC TEST CONDITIONS

i300/I200 COMPLIANT TO LISTED EMC TEST CONDITIONS			
Environment	Test Method Standard	Test Parameters	
Radiated Emissions	ISO 13309:2010	30 MHz to 1000 MHz	
	ISO 13766:2006	BB and NB Scans	
	ISO 14982:1998	Ambient Baseline Before & After	
Bulk Current Injection (BCI)	ISO 11452-4:2011	20 MHz to 400 MHz	
		100 mA, 80% AM at 1kHz	
Radiated Immunity	100 11/052 2:200/	400 MHz to 2000 MHz	
	ISO 11452-2:2004	100 V/m, 80% AM at 1kHz and PM	
Conducted Transients	ISO 7637-2:2011	Pulses 1, 2a, 2b, 3a, 3b at Test Level IV	
	ISO 7637-2:2004	Pulse 4 at Test Level IV	

# **MOUNTING/** INSTALLATION

Do not place this device in an environment with Helium concentrations greater than the normal atmosphere. The helium will permeate the housing and affect sensors. The housing seal allows Helium to enter/leave so that helium does not accumulate. The IMU should not be subjected to contact with any fuels, lubricants, solvents, or their vapors.

The accelerometer and gyro sensors are mounted in a normally aligned, right-handed axis configuration that is nominally aligned with the IMU axes as shown in the figure below. If the X axis is pointed up away from the Earth's surface, the accelerometer reading will be positive.

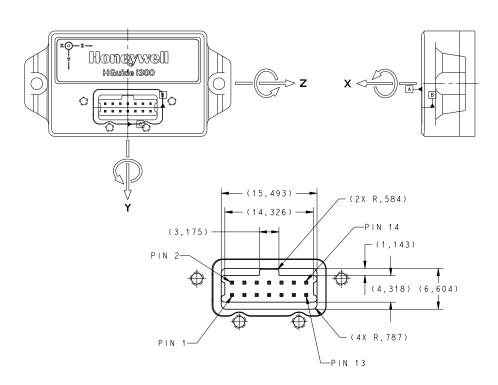
The i300/i200 nominally weighs 35 grams. The packaging is compliant to IP68; however, do not intentionally submerse device under water.

Recommended mating connectors are SAMTECH Part Numbers CLT-107-02-L-D-BE or CLT-107-02-F-D-A or equivalent.

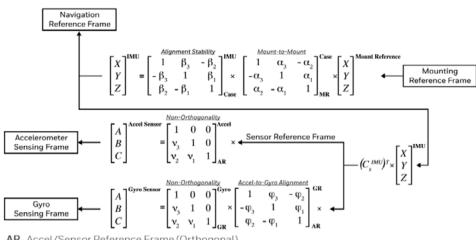
The center of gravity and center of navigation are located at the approximate geometric center. A CAD compatible STP file is available from Honeywell upon request.

#### **ATTENTION:**

IMUs are precision instruments which measure angular rate and linear acceleration across a broad temperature range. Because of their precision, users often interpret real motion (both angular and linear) as sensor noise. This noise can often be coupled mechanically through the mounting plate. Installation on a thin structure is generally not desirable. Placement at anti-nodes will minimize angular rotation and maximize linear displacement. Placement at nodes will maximize angular rotation and minimize linear displacement.



## ALIGNMENT AND ORTHOGONALITY



**AR** Accel/Sensor Reference Frame (Orthogonal)

**GR** Gyro Reference Frame (Orthogonal)

**C**<sup>IMU</sup><sub>s</sub>Gyro Reference Frame (Orthogonal)

ALIGNMENT AND ORTHOGONALITY				
Parameter	Requirement	Units		
Mount to Mount with Pins	3500	µrad max		
Sensor Frame Alignment Stability	<375	μrad 1σ		
Accelerometer Non-orthogonality	<375	μrad 1σ		
Accelerometer to Gyro Alignment	<375	μrad 1σ		
Gyro Non-Orthogonality	<375	μrad 1σ		

#### Alignment / Orthogonality Note

Honeywell navigation system equations implement alignment / orthogonality as shown. These equations are provided to customers for understanding of the parameters provided. Customers may optionally choose to implement these equations into their own navigation equations. Values shown in above table are "as delivered" and include Honeywell factory calibration.



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